

Formulas to remember (1)

$$\odot \vec{v} = \dot{r} \hat{r} + \dot{\theta} r \hat{\theta}$$

$$\odot \vec{a} = (\ddot{r} - \dot{\theta}^2 r) \hat{r} + (2\dot{r}\dot{\theta} + \ddot{\theta} r) \hat{\theta}$$

$$\odot U = - \int_{\vec{r}}^{\vec{r}'} \vec{F} \cdot d\vec{r}$$

$$\odot R_{cm} = \frac{\sum m_i r_i}{\sum m_i}$$

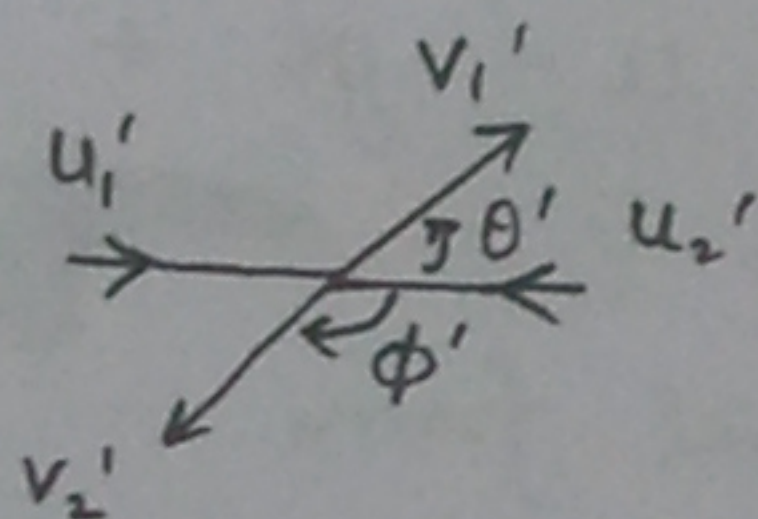
⊙ For an elastic collision as observed from COM frame,

$$|v_1'| = |u_1'|$$

$$|v_2'| = |u_2'|$$

ie. magnitude of velocities do not change from COM frame of reference, only change in direction.

$$\theta' + \phi' = \pi$$



$$\tan \theta = \frac{\sin \theta'}{\cos \theta' + \left(\frac{m_1}{m_2}\right)}$$

$$\phi = \left(\frac{\phi'}{2}\right)$$

⊙ Energy is transformed in the same way other vectors are transformed w.r.t. COM Transformation.

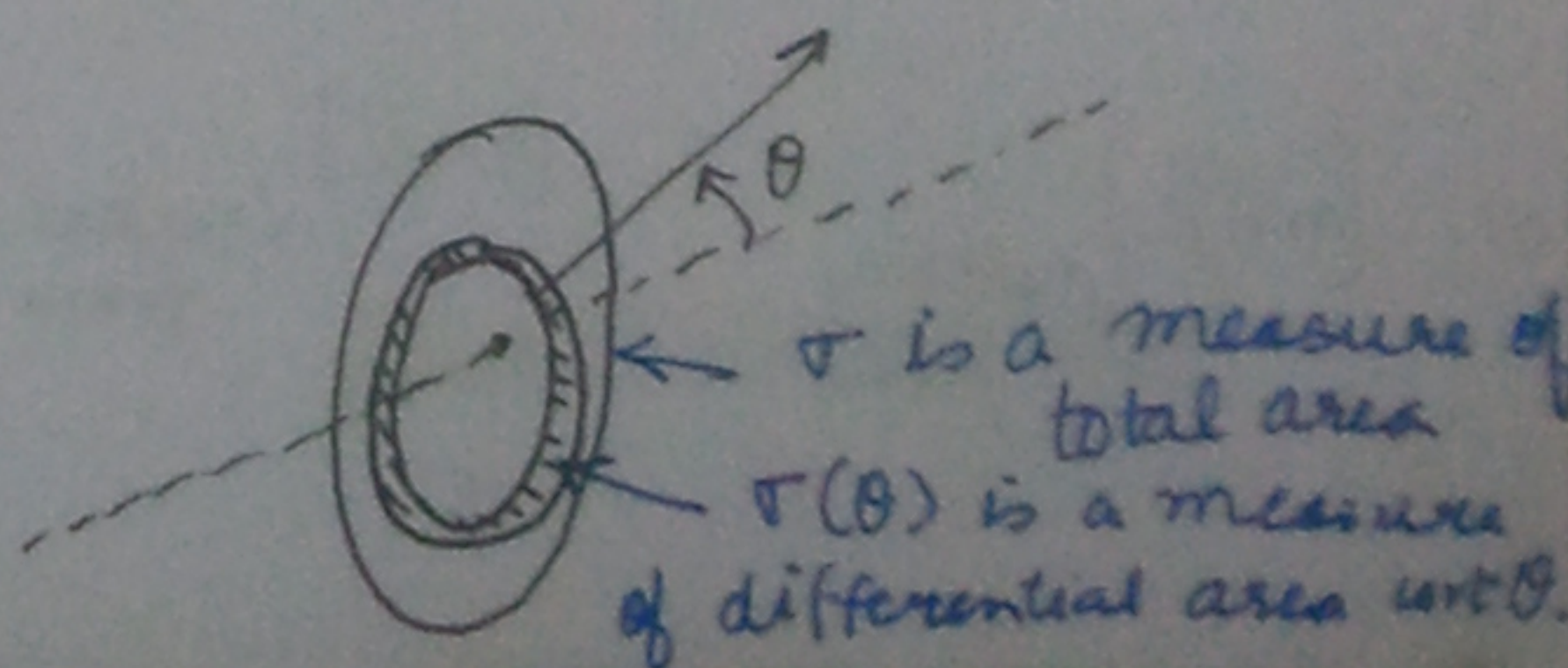
$$E' = E - E_{cm}$$

$$\odot \sigma = \int_{\theta=0}^{\theta=\pi} \sigma(\theta) 2\pi \sin \theta d\theta$$

$$d\Omega = 2\pi \sin \theta d\theta$$

$$\odot \sigma(\Omega) = \left[\frac{dN}{I d\Omega} \right] \quad \text{and} \quad \sigma = \left[\frac{N}{I} \right]$$

$$\odot \sigma(\theta) = -(b/\sin \theta) * (db/d\theta)$$



⊙ For Rutherford scattering,

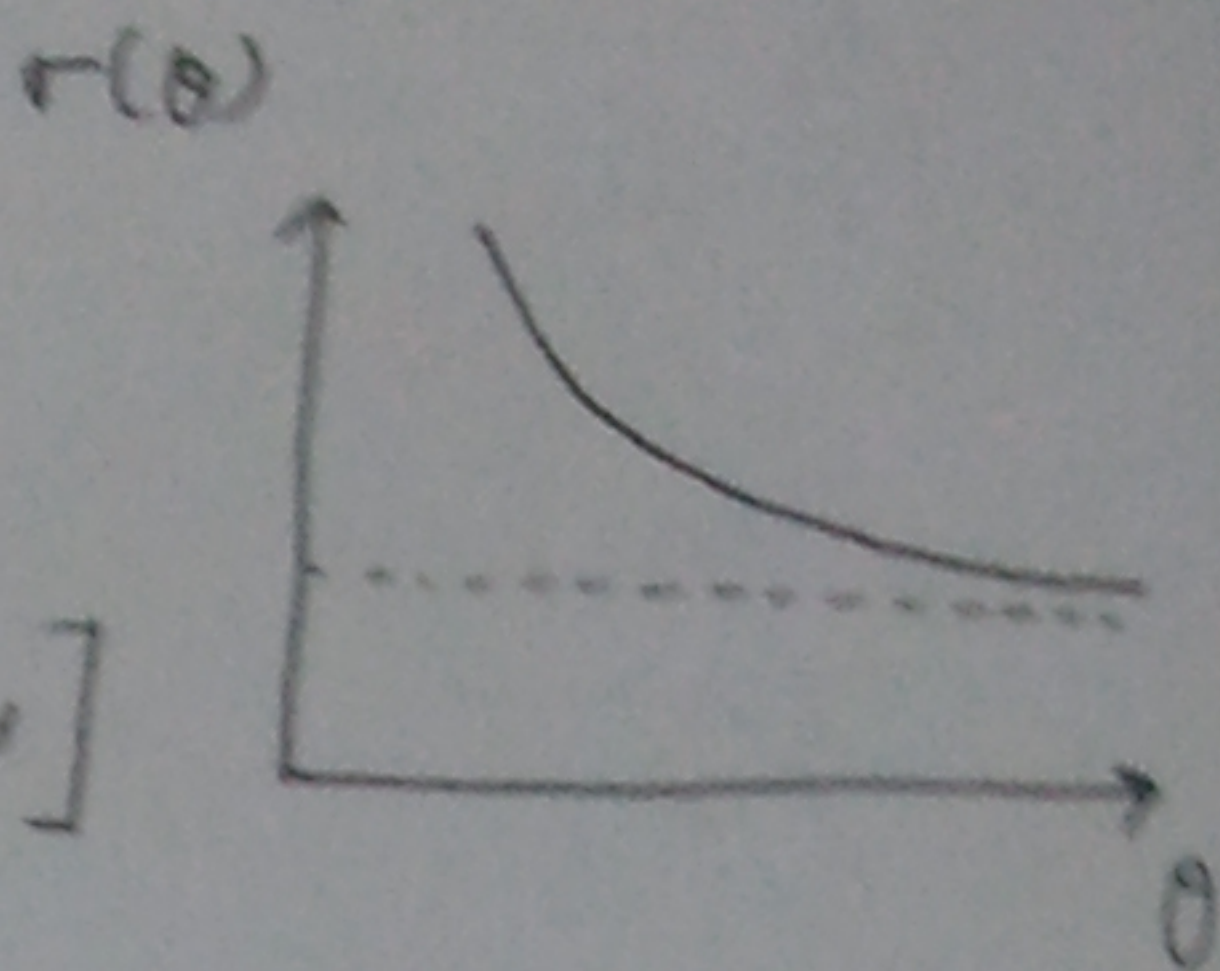
$$b = \frac{C_1}{2E_d} \cot\left(\frac{\theta}{2}\right)$$

where $C_1 = \left[\frac{Z_1 Z_2 e^2}{4\pi\epsilon_0} \right]$ in F

Corresponding


$$\sigma(\theta) = \frac{C_1^2}{16 E_d^2} \operatorname{cosec}^4\left(\frac{\theta}{2}\right)$$

[All Parameters w.r.t. Centre of m_2 or Centre of Mass]



⊙ For hard-sphere scattering

$$\sigma(\theta) = \left[\frac{R^2}{4} \right]; \quad \sigma = \pi R^2$$

⊙ Rocket Motion : $v = u - gt + u_{\text{exhaust}} \ln\left(\frac{M_0}{M}\right)$
 Thrust = $-\left(\frac{dM}{dt}\right) u_{\text{exhaust}}$; 

⊙ For observation in non inertial frame

$$[F]_{\text{observed in non inertial frame}} + F_{\text{pseudo}} = m [a]_{\text{observed in non inertial frame}}$$

$$\vec{a} = \left(\frac{d\vec{p}}{dt} \right) \text{ by definition irrespective of frame of ref}$$

⊙ For rotating vector $\vec{\lambda}$ $\frac{d\vec{\lambda}}{dt} = [\vec{\omega} \times \vec{\lambda}]$

⊙ Transformation from rotating frame to inertial frame,

$$\frac{d\vec{\lambda}}{dt} = \left[\frac{d\vec{\lambda}}{dt} \right]_{\text{observed in rotating frame}} + [\vec{\omega} \times \vec{\lambda}]$$

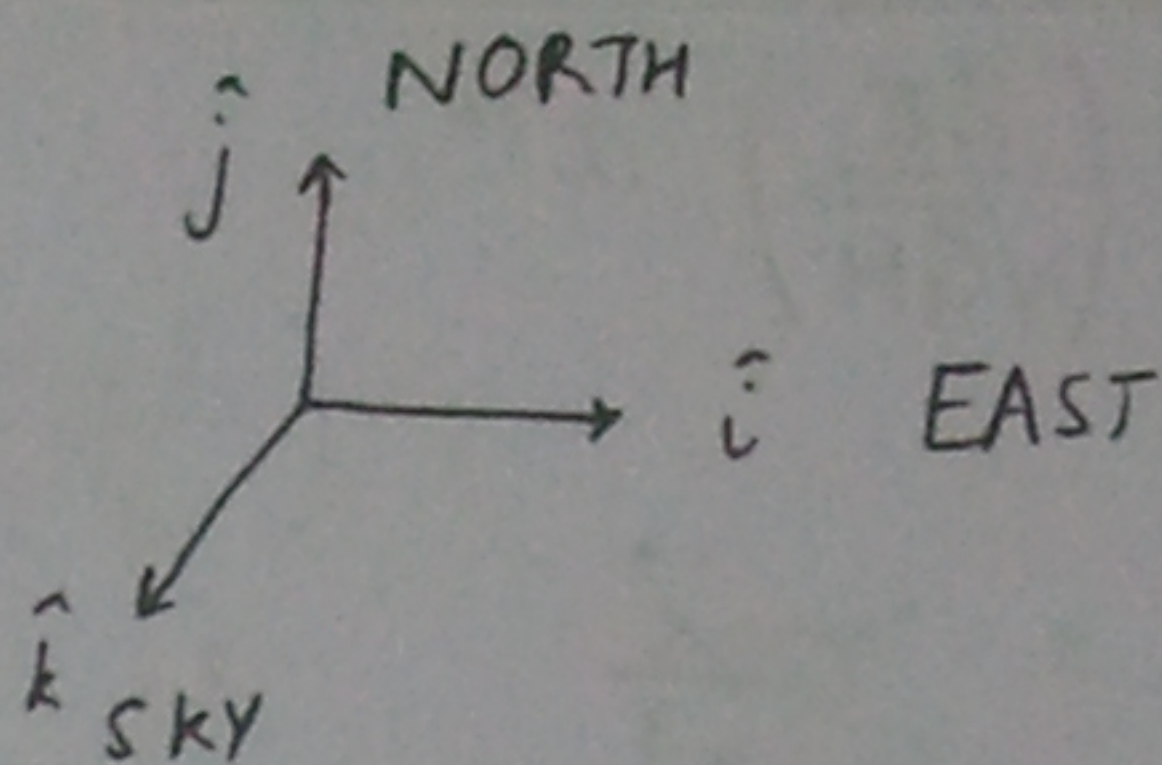
$$[m\vec{a}]_{\text{rotating frame}} = [m\vec{a}]_{\text{inertial frame}} + \underbrace{[-2m(\vec{\omega} \times \vec{v}')]_{\text{Coriolis}}}_{\substack{\uparrow \\ \text{velocity measured in rotating frame}}} + \underbrace{[-m\vec{\omega} \times (\vec{\omega} \times \vec{r}')]_{\text{Centrifugal}}}$$

Formulas to remember (2)

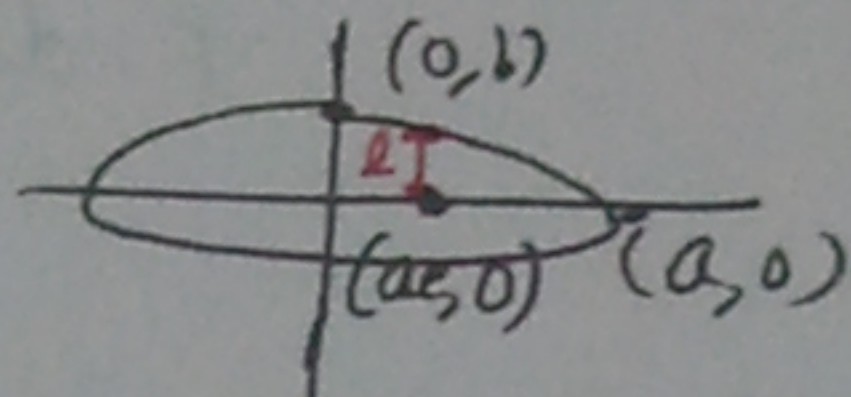
$$\vec{\omega} = \omega \cos \lambda \hat{j} + \omega \sin \lambda \hat{k}$$

$$F_{\text{Coriolis}} = -2m (\vec{\omega} \times \vec{v})$$

$$F_{\text{centrifugal}} = -m \vec{\omega} \times (\vec{\omega} \times \vec{r})$$



Ellipse: $\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$



$$\frac{l}{r} = (1 + e \cos \theta) \quad l = \left(\frac{b^2}{a}\right) = a(1 - e^2)$$

$$b^2 = a^2(1 - e^2)$$

$$r_{\text{max}} = \frac{l}{1 - e} = a(1 + e)$$

$$r_{\text{min}} = \frac{l}{1 + e} = a(1 - e)$$

e Conic
0: ellipse

0 < e < 1: ellipse

1: Parabola

e > 1: Hyperbola

$$\left(\frac{dA}{dt}\right) = \left(\frac{J}{2\mu}\right) = \text{Const.}$$

Trajectory of Motion under Central Force:

$$\frac{d^2 u}{d\theta^2} + u = - \left[\frac{\mu F(u)}{J^2 u^2} \right]$$

always $\frac{J^2}{\mu^1}$

For $F = -\left(\frac{k}{r^2}\right)$, the trajectory is

$$\frac{\left[\frac{J^2}{k\mu}\right]}{r} = 1 + \sqrt{1 + \frac{2EJ^2}{\mu k^2}} \cos(\theta - \theta_0)$$

$$v_{\text{esc}} = \sqrt{\frac{2GM}{r}} \quad [\text{in general}]$$

$$v_0 = \sqrt{\frac{GM}{r}} \quad [\text{for circular case}] \dots \dots$$

$$\nabla \left(\frac{1}{r} \right) = -\frac{\vec{r}}{r^3} \quad ; \quad F = -\nabla U \quad \text{and} \quad E = -\nabla V$$

$$U = -\int_{\infty}^r F \cdot d\vec{r} \quad \text{and} \quad V = -\int_{\infty}^r E \cdot d\vec{r}$$

$$\vec{F} \cdot d\vec{s} = -dU \quad \text{and} \quad \vec{E} \cdot d\vec{e} = -dV$$

Gravitational self Energy for a sphere = $-\frac{3}{5} \left(\frac{GM^2}{R} \right)$

Gauss law : $\oint_s E \cdot ds = (-4\pi G) \text{ enclosed}$

or

$$\nabla \cdot \vec{E} = -4\pi \rho G$$

$$\vec{J} = \sum_i \vec{r}_i \times \vec{p}_i$$

$$\begin{bmatrix} J_x \\ J_y \\ J_z \end{bmatrix} = \begin{bmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} \end{bmatrix} \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix}$$

$$I_{xx} = \sum m_i (x_i^2 + y_i^2)$$

$$I_{xy} = -\sum m_i (x_i y_i)$$

$$I_{xz} = -\sum m_i (x_i z_i)$$

$$K.E. = \frac{1}{2} \vec{\omega} \cdot \vec{J} = \frac{1}{2} \vec{\omega} \cdot [\vec{I} \vec{\omega}]$$

Torque free Motion of a symmetric body :

$$\rightarrow \frac{\omega_1^2}{\left(\frac{2k}{I_1}\right)} + \frac{\omega_2^2}{\left(\frac{2k}{I_2}\right)} + \frac{\omega_3^2}{\left(\frac{2k}{I_3}\right)} = 1$$

POINSON EQUATION

$$\rightarrow \vec{\omega} = A \sin \Omega t \hat{i} + A \cos \Omega t \hat{j} + \omega_3 \hat{k}$$

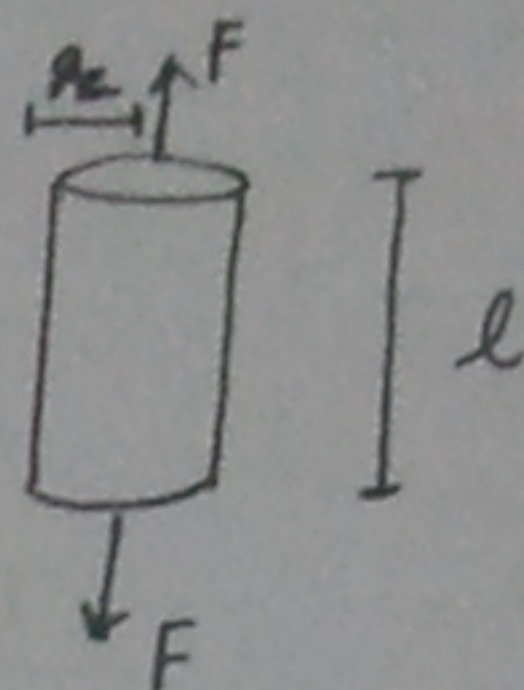
where $\Omega = \omega_3 \left(1 - \frac{I_2}{I_1} \right)$
[Precession Velocity]

Formulas to remember (3)

Precession Motion of a Top:

$$\text{Precessional Angular Velocity } \Omega = \left(\frac{mgl}{I \omega} \right)$$

Lateral Change in length $da = \left(\frac{\sigma P}{Y} \right) a$



$$Y = 3K(1 - 2\sigma)$$

$$Y = 2\eta(1 + \sigma)$$

$$\eta = \frac{(F/A)}{(dv/dz)}$$

Stoke's law: $F = 6\pi\eta r v$: Frictional drag on a spherical body

Terminal velocity = $\frac{2r^2g(\rho - \sigma)}{9\eta}$

$R = \left(\frac{\rho v D}{\eta} \right)$ Critical value: 1000

Poiseuille's Formula for flow in Capillaries $Q = \left(\frac{\pi P a^4}{8\eta l} \right)$

Bernoulli Theorem: $\frac{1}{2}\rho v^2 + P + \rho gh = \text{const.}$

Venturimeter: $Q = C\sqrt{h}$ where $C \propto \sqrt{\frac{A_1^2 A_2^2}{A_1^2 - A_2^2}}$

Lorentz Transform: $x' = \alpha(x - vt)$
 $t' = \alpha\left(t - \frac{v}{c^2}x\right)$

Length Contraction $\Rightarrow L = \left(\frac{L_0}{\alpha} \right) \Rightarrow L_0$: Proper length is largest

Time Dilation $\Rightarrow T = \alpha T_0 \Rightarrow T_0$: Proper Time is least

⊙ Composition of Velocities

$$u_x' = \frac{u_x - v}{1 - u_x \frac{v}{c^2}}$$

$$u_y' = \frac{u_y}{\alpha \left(1 - u_x \frac{v}{c^2}\right)}$$

$$u_z' = \frac{u_z}{\alpha \left(1 - u_x \frac{v}{c^2}\right)}$$

⊙ Space-Time Coordinates: (x, y, z, t)

⊙ Minkowsky's 4-d Coordinates

$$(x_1, x_2, x_3, x_4)$$

where

$$x_1 = x$$

$$x_2 = y$$

$$x_3 = z$$

$$x_4 = ict$$

[Also called 4-vector]

⊙ Space Time Interval is defined as interval in Minkowsky's Coordinates i.e.

$$(x_1 - x_2)^2 + (y_1 - y_2)^2 + (z_1 - z_2)^2 + i^2 (t_1 - t_2)^2$$

$$= (dx)^2 + (dy)^2 + (dz)^2 - c^2 dt^2$$

⊙ Velocity 4-vector or Four Velocity = $\begin{bmatrix} \alpha v_x \\ \alpha v_y \\ \alpha v_z \\ i \alpha c \end{bmatrix}$

Momentum 4-vector or Four Momentum = $m_0 \begin{bmatrix} \text{velocity} \\ \text{Four vector} \end{bmatrix}$

$$= \begin{bmatrix} \alpha m_0 v_x \\ \alpha m_0 v_y \\ \alpha m_0 v_z \\ i \alpha m_0 c \end{bmatrix} = \begin{bmatrix} m v_x \\ m v_y \\ m v_z \\ i m c \end{bmatrix} = \begin{bmatrix} p_x \\ p_y \\ p_z \\ i \frac{E}{c} \end{bmatrix}$$

⊙ Minkowsky's Transformation Matrix: $\begin{bmatrix} \alpha & 0 & 0 & i \alpha \beta \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ -i \alpha \beta & 0 & 0 & \alpha \end{bmatrix}$

⊙ $m' = \frac{m}{\alpha} \frac{\left(1 - \frac{u_x v}{c^2}\right)}{\sqrt{1 - \frac{v^2}{c^2}}}$

$$p_x' = \left(\frac{p_x - \frac{E v}{c^2}}{\sqrt{1 - \frac{v^2}{c^2}}} \right)$$

$$E' = \left(\frac{E - v p_x}{\sqrt{1 - \frac{v^2}{c^2}}} \right)$$

since $x = ct$ & $E = mc^2$
 $x \leftrightarrow E$, $t \leftrightarrow p$
 similar transformation as in Lorentz or Minkowsky

Formulas to remember (4)

$$E = mc^2$$

$$1 \text{ a.m.u} : 931 \text{ MeV}$$

$$e^- : 0.51 \text{ MeV}$$

$$\text{Proton} : 938 \text{ MeV}$$

$$\text{Neutron} : 940 \text{ MeV}$$

Rest Mass Energies
 $= m_0 c^2$

NEUTRON is
heaviest

$$E^2 = (pc)^2 + (m_0 c^2)^2$$

$$\gamma' = \gamma \sqrt{\frac{1 - \frac{v}{c}}{1 + \frac{v}{c}}}$$

$$\text{or } \gamma' = \gamma \sqrt{\frac{1 + \frac{v}{c}}{1 - \frac{v}{c}}}$$

$$f_0' = \alpha^2 f_0$$

Aberration :

$$\tan \theta = \left(\frac{\tan \theta'}{1 + \frac{v}{c \cos \theta'}} \right)$$

Rocket Motion

$$M = M_0 - \alpha t$$

$$M = M_0 (1 - \beta t)$$